

Error Propagation in Satellite Multi-image Geometry

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Abstract— This paper describes an investigation of the source of geospatial error in digital surface models (DSMs) constructed from multiple satellite images. In this study the uncertainty in surface geometry is separated into two spatial components; global error that affects the absolute position of the surface, and local error that varies from surface point to surface point. The global error component is caused by inaccuracy in the satellite imaging process, mainly due to uncertainty in the satellite position and orientation (pose) during image collection. The key sources of local error are; lack of surface appearance texture, shadows and occlusion. These conditions prevent successful matches between corresponding points in the images of a stereo pair.

A key result of the investigation is a new algorithm for determining the absolute geo-position of the DSM that reflects the pose covariance of each satellite during image collection. This covariance information is used to weigh the evidence from each image in the computation of the global position of the DSM. The use of covariance information significantly decreases the overall uncertainty in global position and results in a 3-d covariance matrix for the global accuracy of the DSM. This covariance matrix defines a confidence ellipsoid within which the actual error must reside. Moreover, the absolute geo-position of each image is refined to the reduced uncertainty derived from the weighted evidence from the entire image set.

The paper also describes an approach to the prediction of local error in the DSM surface. The observed variance in surface position within a single stereo surface reconstruction defines the local horizontal error. The variance in the fused set of elevations from multiple stereo pairs at a single DSM location defines the local vertical error.

These accuracy predictions are compared to ground truth provided by LiDAR scans of the same geographic region of interest. The prediction of global and local error is compared to the actual errors for several geographic locations and mixes of satellite type. The predicted error bounds contain the observed errors according to the allowed percentage of outliers.

Index Terms—Stereo, Photogrammetry, Digital Surface Models, Satellite Imaging

I. INTRODUCTION

This paper is focused on the problem of predicting the expected geo-spatial accuracy of digital surface models (DSMs) generated from general satellite image archives. Recent advances in stereo reconstruction algorithms have

enabled the generation of a single DSM from hundreds of stereo pairs, as are available over many locations of interest on the Earth, providing complementary and redundant elevation samples. Satellite image datasets are available as a service to commercial vendors of such DSMs [1] such as Vricon and provided as open challenge datasets for academic research [2]. Moreover, Department of Defense contractors have unlimited free use of MAXAR's commercial archive of millions of satellite images [3]. The MAXAR archive provides at least 10-20 images for most regions on the Earth landmass with many areas having coverage of 100s of images.

Commercial satellite images are equipped with metadata that describes the projection from 3-d geographic coordinates to image coordinates in the form of a rational polynomial model (RPC). Given the covariance of satellite pose errors during image collection, the DSM geo-positioning accuracy can be analyzed by a process known as *error propagation* [4] where known error covariances are passed through the algorithmic stages of stereo geometry formation to predict the full 3×3 covariance matrix associated with the global position of the DSM. In this paper, traditional error propagation is augmented with the direct computation of global DSM position without requiring an initial guess or multiple iterations.

Formation of multi-view stereo surface geometry is a complex process that depends on many factors including the fusion of 100s of 3-d points associated with each DSM grid cell, where each associated pointset can contain a high proportion of outliers (see Fig. 1). Consequently, traditional error propagation methods cannot be applied to determine the local elevation error at each grid cell. This paper introduces new methods to estimate predicted accuracy in multi-image DSM products and demonstrates these methods on a heterogeneous mix of satellite platforms and for various scene contexts and geographic locations.

A. Digital Surface Models from Satellite Images

The rational polynomial coefficient (RPC) projection metadata provides the necessary information to produce a rectified stereo pair from two arbitrary commercial satellite images [5]. The result is that vast archives of satellite image data can be exploited to produce DSMs for many regions on the Earth surface and over an extensive range of time

Submitted April 10, 2021, final revision submitted September 2, 2021.

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intervals. Moreover, images from different satellite platforms can be combined into a stereo pair further extending the population of images that can be used to generate surface models.

B. Semi-global Matching

Another key advance that makes the generation of accurate and high-resolution DSMs from general image archives feasible was the development of the semi-global matching (SGM) stereo algorithm by Hirschmüller [6]. The SGM algorithm provides a dense reconstruction of 3-d geometry based on matching corresponding pixels in the left and right stereo images using a dynamic programming scheme. The SGM algorithm makes a global assessment of the cost for assigning corresponding pixels and thus errors in matching left and right image pixels are greatly reduced, while at the same time maintaining computational efficiency. The match cost is based on neighborhood appearance measures at each pixel. However, matching errors are unavoidable in the case of occlusion, shadowing and actual scene content differences, such as seasonal changes.

C. Geo-registration

To produce consistent geometry from multiple stereo pairs it is necessary to geo-register the set of input images into a common spatial coordinate frame with a high degree of accuracy. The registration must be achieved with sub-pixel relative accuracy so that a 3-d point produced by one stereo pair is closely aligned with the same point produced by other pairs. It is necessary to carry out the alignment of all the images in the dataset at the same time to achieve consistent geometry in a procedure called bundle adjustment [7]. The bundle adjustment algorithm relies on corresponding image features to provide constraints to solve for the unknown alignment transformation of each image. A set of image correspondences across a subset of the input images corresponding to a single 3-d point is called a *track*. Bundle adjustment simultaneously determines the 3-d point coordinates and adjusts the projection models of the track images to achieve accurate relative co-registration. Note that accurate relative registration does not guarantee accurate global registration.

D. Multiple Stereo Pair Fusion

Finally, it is necessary to fuse together multiple sets of densely reconstructed 3-d points, generated from their respective multiple stereo pairs, to form a complete surface. No single pair can produce a complete surface model given the occlusion present in a 3-d scene. For example, in urban environments many viewpoints are required to observe all surfaces due to occlusion by tall buildings. Similarly, cast shadows in the scene observed by a single stereo pair produce areas of unknown geometry due to the failure to find matching appearance values between a shadowed and visible surface point. Shadow differences occur with images taken at different time of the year or with different satellites. Hirschmüller suggested some fusion strategies in his original paper, and the topic has received much attention in the literature [8,9]. For the case of a digital surface model, multiple stereo pairs produce a set of elevation values for each position in the x, y

plane and fusion is the process of selecting a single elevation that represents the set. One common approach is to find the median elevation. However, the median estimate can produce erroneous results in the case of high outlier populations. A more robust fusion method is introduced in section VI.C.

II. TYPES OF DIGITAL SURFACE MODEL ERRORS

A DSM is a function, $z(x, y)$ where a horizontal position, (x, y) , maps to a single elevation value, z . The DSM can be represented in different coordinate systems such as geographic coordinates based on WGS84, where x and y are in degrees and for Universal Transverse Mercator (UTM) in meters. DSMs are often encoded as GEOTIFF files with metadata that defines the geographic coordinate system and DSM grid spacing.

A. Horizontal Error

Ideally, the (i, j) location of each DSM pixel corresponds to the (x, y) location of that surface position in the actual scene. The location will be quantized by the granularity of the DSM image pixel spacing. However, the error in the horizontal location can be much larger than the pixel spacing. Denote the actual ground truth location of pixel (i, j) as (x_{gt}, y_{gt}) and denote the DSM position as specified by a GEOTIFF image as (x_{dsm}, y_{dsm}) . The (x, y) error is defined as,

$$\epsilon_{Horizontal} = (x_{dsm} - x_{gt})\hat{x} + (y_{dsm} - y_{gt})\hat{y} \quad (1)$$

typically called *horizontal* error since the error vector lies in the horizontal x, y plane.

B. Vertical Error

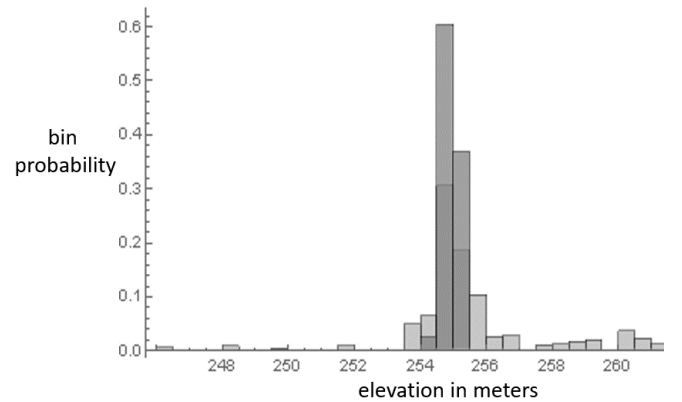


Fig. 1 The distribution of 100 stereo-derived elevation values (light grey) for a single (x, y) location. The dark grey histogram bins represent the chosen consensus set of inliers. (see section VI.C)

Even if the horizontal position is perfectly aligned with its ground truth position there can be error in the elevation, $z(x, y)$, specified by the DSM at each (x, y) location. One component of vertical error is the result of fusing 100 or more stereo pair estimates for $z(x, y)$. There can be many outliers due to errors in stereo matching as shown in Fig. 1. The fused result is thus perturbed from the ground truth value even with minimum variance estimators. Moreover, each stereo pair may not be in exact geo-registration and so there can be blurring of surface features due to fusing the displaced surfaces.

C. Global Error

Another component of horizontal and vertical error is due to a global transformation of the entire DSM with respect to the correct geographic location of the surface. Such global transformations are caused by error in the image geo-registration process. The geo-registration algorithm does not make use of any known ground points but relies on the bundle-adjustment algorithm to average out errors in the metadata that defines the projection of a 3-d point into image coordinates. A large component of this type of error is due to uncertainty in the pointing direction of the satellite during image collection. The satellite is hundreds of kilometers from the Earth surface and so even a few micro-radians of orientation error can cause meters of positional error at the surface. If the satellite metadata is biased, then a residual global 3-d error in the position of the entire DSM will occur.

D. Local Error

Horizontal and vertical errors can vary from pixel to pixel and become larger near the boundaries of building roofs or other sharp changes in surface elevation. These larger errors are due mainly to error in pixel correspondences within the SGM algorithm due to variations in appearance between images so that errors in stereo correspondence occur. For example, some locations can be missed entirely if one image in the stereo pair has a surface point in shadow and the other does not.

Error can also arise due to the quantization of corresponding locations in a stereo pair since SGM match locations are in terms of integer pixel coordinates, i.e., ± 1 pixel in rectified image space, and depending on the view angle separation of the pair, this quantization can translate to a significant error in the 3-d position of the intersection point. That is, the geometric accuracy of the intersection of two rays degrades as the rays become more parallel.

Another source of horizontal error is the assignment of 3-d points generated from the stereo pairs to quantized DSM grid locations. The 3-d points assigned to a pixel bin form the population that is used to fuse the z value for that bin. However, the centroid of the population is not generally at the center of the pixel and so a shift in horizontal location for the fused z value is generated.

E. Previous approaches

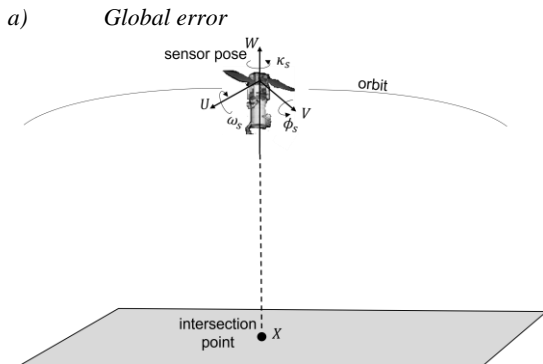


Fig. 2 The imaging satellite pose covariance coordinate system. The covariance of intersection point X is derived from the known covariance of sensor pose.

A standard approach to model global error of multi-image geo-positioning (MIG) such as a DSM is to propagate known error covariance in the pose (position and attitude) of an imaging satellite to the 3×3 covariance matrix of 3-d point coordinates [10]. The coordinate system for satellite position and attitude (pose) error is illustrated in Fig. 2. The analysis for the covariance of X for n satellite images requires three matrices: Σ_p the $6n \times 6n$ covariance matrix for satellite pose errors; B_p the $2n \times 6n$ Jacobian matrix of derivatives of image coordinates with respect to pose; and B the $2n \times 3$ matrix of derivatives of image coordinates with respect to the coordinates of X .

Denote the error in the intersection point as ∂X so that $X = X_0 + \partial X$. The covariance matrix for ∂X is denoted as $P = \langle \partial X \partial X^T \rangle$, where $\langle \cdot \rangle$ indicates expectation over an ensemble of random instances. Given these definitions the error propagation proceeds as follows.

$$W = (B_p \Sigma_p B_p^T)^{-1} \quad (2)$$

$$P = (B^T W B)^{-1} \quad (3)$$

The $2n \times 2n$ weight matrix W is the inverse covariance matrix for errors in image coordinates due to satellite pose errors. This weight matrix propagates pose errors to produce the covariance matrix P for intersection point coordinates via the matrix B . This analysis in terms of derivatives is valid if the intersection point X is near its true value and the pose errors are small enough so that first derivatives of the projection from 3-d to 2-d are sufficiently accurate.

There are two limitations of this approach: 1) it is necessary to have an initial guess for the intersection point; 2) there is no prescription for the application of satellite pose uncertainty to the computation of the initial guess for the intersection point. Instead, a different procedure to produce the initial guess is recommended such as least-squares line intersection [11]. The method does however provide a way of refining the initial guess based on the pose covariance information. The value of X can be iteratively updated by,

$$\partial X = P B^T W (\tilde{x} - x), \quad (4)$$

where x is the image projection of X and \tilde{x} is the declared image position of X .

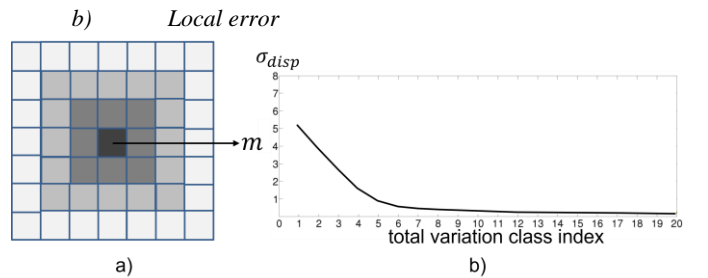


Fig. 3 Estimating local errors in disparity. a) The set of neighborhood rings for computing total variation class. b) The relation between total variation class and the standard deviation of disparity.

There have been previous efforts to model the effect of the types of local errors defined above. A method called *total variation* can be applied to the disparity image generated by the SGM algorithm. The disparity image encodes the row by row pixel shift (disparity) required to match each pixel in the one stereo pair image to a pixel on the same row in the other image. It has been proposed by Kuhn that localized, large

changes in disparity will be characteristic of high uncertainty in the reconstructed 3-d geometry [12]. Kuhn calls the quantification of these rapid changes total variation.

The computation of total variation is illustrated in Fig. 3a). The class index of total variation of disparity is computed as,

$$TV_{class} = \max_n \left(\sum_{m=1}^n \frac{1}{8m} \sum_{i,j \in N_m} \sqrt{|d_{i+1,j} - d_{i,j}| + |d_{i,j+1} - d_{i,j}|} \right) < \theta \quad (5)$$

, where m is the radius of a neighborhood of pixels, N_m as shown in Fig. 3a), and $d_{i,j}$ is the disparity at neighborhood location (i, j) . If the disparity image is smooth, a large radius is required before changes in disparity occur that exceed the threshold θ . Conversely, if the disparity region is noisy then the threshold will be exceeded for small values of m . The index value can be mapped to the corresponding value of actual standard deviation in disparity by observing the values of TV_{class} on training data equipped with ground truth disparity values. An example curve relating TV_{class} to the standard deviation of disparity, σ_{disp} , is shown in Fig. 3 b). The standard deviation in disparity is then used to compute the covariance matrix of the reconstructed 3-d points for the stereo pair.

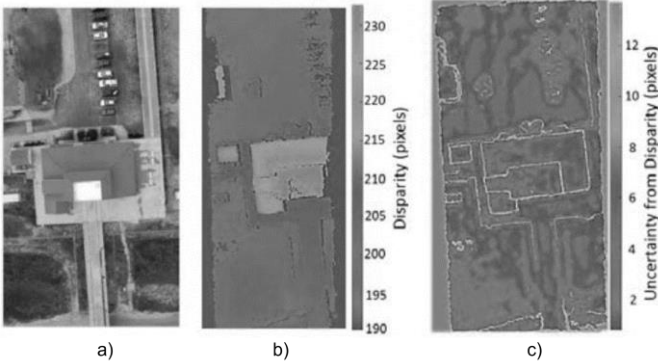


Fig. 4 a) Drone image. b) Disparity image. c) Predicted uncertainty in disparity. From Rodarmel *et al* [13].

This approach has been applied to estimating the local errors in 3-d point clouds derived from aerial images collected by a small drone. The work by Rodarmel *et al.* [13] applies rigorous error propagation methods to predict spatially varying point covariance. An example of predicted disparity standard deviations from the disparity image, see Fig. 4c), is shown to bound the actual errors as determined from ground truth point positions.

F. Proposed approach

A new approach that is keyed to the specific case of DSM construction from satellite images is proposed. In the development to follow, a typical multi-image stereo DSM reconstruction system is used to illustrate the method and study the results of applying satellite pose covariance to the prediction of local and global errors manifested for a given mix of satellite sensor error characteristics. The DSM reconstruction computational pipeline is formulated in terms of services such as radiometric correction, region of interest tiling, image geo-registration, stereo-pair processing and fusion, hereafter known as the “system.” The following two

sections address “global” and “local” errors, respectively. The former is driven by intersection geometry and impacts all points in the scene in the same manner, while the latter is driven by the per-pixel matching uncertainty to reconstruct the disparities, or scene depths.

a) Global error

The prediction of global accuracy is based on propagating the pose covariance of the satellite sensors throughout the geo-registration process and its construction of the 3-d intersection point for matched feature correspondences across multiple images (called a track). Such tracks are found by matching features such as SIFT [14] between pairs of images and recursively building up matches among a set of images. Proposed matches are tested with respect to the epipolar constraint, i.e., given a correspondence pair, the position of one match must lie close to the epipolar line of the other match.

The SIFT implementation used in the experiments below achieves sub-pixel accuracy by interpolating the maximum value of the Laplacian signal. The largest feature track typically corresponds to SIFT matches from a small dark depression such as a hole, so that illumination and viewpoint variations do not affect the match. The track feature correspondences enable the simultaneous determination of the 3-d point formed by the intersection of rays back-projected from each feature location and the image translations required to co-register the 3-d to 2-d projection functions for the track images.

In this approach, the satellite pose covariance informs the determination of the 3-d track point by weighing the accuracy of sensor rays cast from each track correspondence. The ray intersection algorithm uses covariance-weighted least squares to find the 3-d point that minimizes the sum of perpendicular distances from the rays to the point. The predicted covariance of the 3-d track point emerges as an integral result of the intersection algorithm. Global DSM placement is rigidly attached to the 3-d track point since the images are all co-registered with respect to the forward projection of the point into each image. The covariance of the global DSM position then corresponds to the covariance in the location of the 3-d track point. Since the solution is based on linear least squares, the result is obtained in closed form not requiring an iterative solution with the attendant risk of non-convergence which may occur when one or more measurements contains a blunder. Thus, outlier detection and removal can occur without the risk of non-convergence.

b) Local error

The prediction of local vertical and horizontal accuracy is based on an analysis of the fusion of stereo point clouds. A stereo pair produces a dense point cloud with a point generated for each valid location in the disparity map. A typical stereo pair will generate millions of points, some of which are invalid due to incorrect correspondences found by the semi-global matching (SGM) algorithm. In the proposed approach the SGM algorithm is run twice, a second time with the images processed in reverse order. The 3-d points generated by the forward and reverse order of the stereo pair

are compared and a high probability assigned to a 3-d point if its forward and reverse locations are close to each other and lower probabilities as the distance between the forward and reverse locations increases. The 3-d point positions for the two pair orders can be considered independent measurements for large errors (blunders) since the triangulated points involve different pixel locations for the forward and reverse orders.

Unlike the comparison of disparity values, geometric distance comparison is invariant to stereo pair view separation and so the point probability values for different pairs can be compared. The probabilities are used to weigh the evidence from multiple stereo pairs in forming the pixel bin populations and in determining the elevation value for each bin. This weighted geometric information can then be used to compute the horizontal and vertical variances for each DSM pixel location, thus taking into account the disparity errors of the SGM algorithm and error in 3-d location based on disparity. Typically, 100s of pairs are combined to form the final DSM surface but as few as three pointsets are sufficient to compute horizontal and vertical variances, albeit with low accuracy. Accounting for the probabilities associated with 3D pointsets from each pair sets this approach apart from a simple unweighted statistical error estimation based upon the scatter of the 3-d points.

The approach just described provides for local and global error prediction and presents a unified approach based on the geometric uncertainty of rays cast from each sensor pixel into 3-d space. The formulation in terms of rays is agnostic to the type of sensor and so the analysis can apply to a wide range of overhead imaging systems including complex wide area motion image (WAMI) scanners and single perspective focal plane cameras as well as a heterogeneous constellation of satellite imaging systems, which is the focus of this paper.

III. THE RAY MODEL FOR SATELLITE IMAGES

A. Geometric Rays

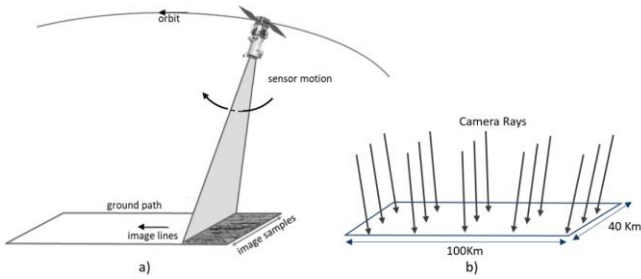


Fig. 5 The formation of a satellite image. a) The image is built up line-by-line. b) The collection of ray cast from image pixels as seen at the corresponding ground points.

The formation of a satellite image is shown in Fig. 5a). The satellite collects the sequence of image lines through a combination of orbital and satellite scanning motion, moving from one line to the next. Note that sensor motion is perpendicular to projection of image lines on the Earth surface and each line is captured by an array of detectors on the focal plane. The projection from ground space to image space is determined from the position and attitude of the spacecraft for each line of the image, as well as the optical system geometry. Any error in the determination of satellite pose will result in an error in the geographic location of the camera rays and thus

in the location of the reconstructed geometry from multiple satellite images. Given a satellite orbit 500Km above the Earth's surface, small attitude errors manifest themselves as translations of the image with respect to the actual ground location. Correction of these pointing errors during geo-registration can be achieved by a 2-d translation of the image.

As shown in Fig. 5b), a typical commercial satellite image spans a region that is 10s of kilometers in extent. An example set of rays is shown in the figure where the ray directions are locally nearly parallel but gradually change direction as the satellite moves along its orbit. The ray geometry, origin and direction, can be derived directly from the third order rational polynomial coefficient (RPC) metadata by computing the back-projection of a pixel location onto a pair of 3-d planes displaced in elevation. The resulting pair of points defines the ray geometry. However, iterative non-linear back-projection computation is costly if carried out at every pixel in the image.

Computation is greatly simplified for many operations if a linear approximation to the projection function is used. The linear approximation is called the affine camera [15] where the ray directions over a small region on the Earth surface are effectively the same, i.e., affine camera rays are parallel. This approximation is very accurate for a typical tile region of 350×350 m., where the change in ray direction over the tile for the World View 3 satellite with an altitude of 650Km is less than 1 milliradian. Thus, even though the projection onto a line of the image sensor is a perspective projection, the projection center is effectively at infinity, as in the affine case. In areas of high relief, tile size is reduced to limit the elevation change within the tile due to memory limitations of the SGM algorithm. Analysis in [16] showed accuracy degradation of up to 0.3 meters at checkpoints for IKONOS stereo extraction in areas spanning 11×11 km; i.e., about 30 times larger in both length and width of the tiles used in this study.

B. The affine camera

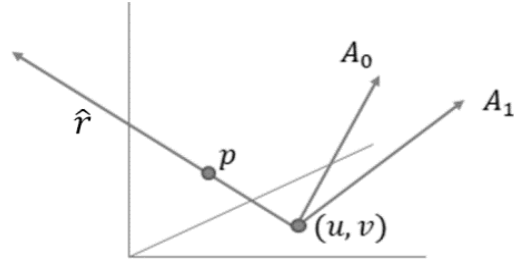


Fig. 6 The ray \mathbf{r} defined by the image location (u, v) for an affine camera. $\hat{\mathbf{r}}$ is the unit ray direction vector and \mathbf{p} is an arbitrary point on the ray. \mathbf{A}_0 and \mathbf{A}_1 are 3-d row vectors from the upper 2×3 sub-matrix of the affine camera matrix. The ray direction is defined by $\mathbf{A}_0 \times \mathbf{A}_1$.

The RPC metadata can be used to construct the affine camera parameters for a given tile by randomly generating a set of 3-d to 2-d correspondences using the forward projection of the RPC model. The affine camera parameters are then estimated from the correspondence set using a linear regression algorithm. The affine camera projection is defined by a 3×4 matrix,

$$C_{aff} = \begin{bmatrix} a_{00} & a_{01} & a_{02} & a_{03} \\ a_{10} & a_{11} & a_{12} & a_{13} \\ 0 & 0 & 0 & 1 \end{bmatrix} = \begin{bmatrix} A_0 & a_{03} \\ A_1 & a_{13} \\ 0 & 1 \end{bmatrix} \quad (6)$$

, so that

IV. SATELLITE POSE COVARIANCE

A. Satellite coordinate systems

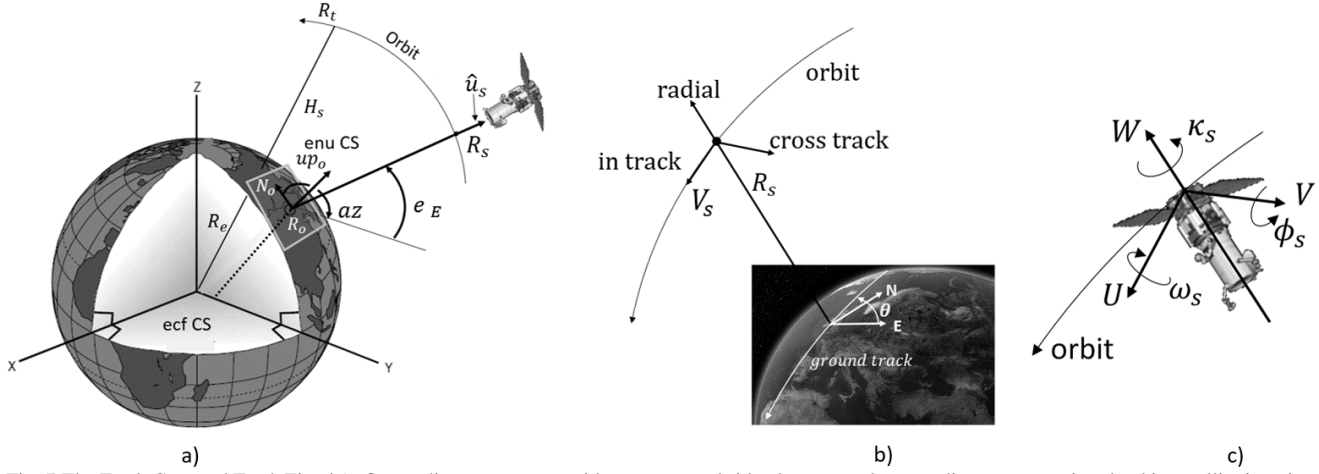


Fig. 7 The Earth Centered Earth Fixed (ecf) coordinate system provides a common bridge between other coordinate systems involved in satellite imaging. The local Cartesian system for the area of interest is in the East North Up coordinate frame, where North and East lie in the ellipsoid tangent plane at R_o and up_o is perpendicular to the ellipsoid at R_o . b) The in track, cross track and radial (ICR) coordinate system defines the position and velocity vector of the satellite's orbital motion. c) The sensor coordinate system, U, V, W is defined by the orientation of the satellite during image acquisition. This coordinate system differs from the ICR system since the satellite can rotate independently of orbital motion.

$$\begin{bmatrix} u \\ v \\ 1 \end{bmatrix} = \begin{bmatrix} A_0 & a_{03} \\ A_1 & a_{13} \\ 0 & 1 \end{bmatrix} \begin{bmatrix} X \\ Y \\ Z \end{bmatrix} \text{ or,} \quad (7)$$

$$\begin{bmatrix} u \\ v \end{bmatrix} = \begin{bmatrix} A_0 \\ A_1 \end{bmatrix} \begin{bmatrix} X \\ Y \\ Z \end{bmatrix} + \begin{bmatrix} a_{03} \\ a_{13} \end{bmatrix} \quad (8)$$

As shown in Fig. 6, the ray direction \hat{r} is given by the cross product of A_0 and A_1 treated as vectors. The direction \hat{r} is the same for any pixel in the image region for which the affine camera is valid. The definition of the ray is completed by specification of any point p along the ray as shown in Fig. 6. Given an image location (u, v) , the point is given by $p = \beta_0 A_0 + \beta_1 A_1$, where

$$\begin{bmatrix} \beta_0 \\ \beta_1 \end{bmatrix} = \begin{bmatrix} A_0 \cdot A_0 & A_0 \cdot A_1 \\ A_1 \cdot A_0 & A_1 \cdot A_1 \end{bmatrix}^{-1} \begin{bmatrix} u - a_{03} \\ v - a_{13} \end{bmatrix} \quad (9)$$

These matrix operations can be carried out with high computational efficiency within a tile, e.g., the matrix inverse in Eq. (9) is the same for every pixel (u, v) in the image region over the tile and so only needs to be computed once.

The affine camera model enables non-iterative low-cost linear algorithms for operations such as: 3-d point triangulation over stereo pairs, correspondence filtering using the epipolar constraint and covariance-weighted least squares estimation of global geo-position of the DSM. It is noted that with respect to global position estimation, the adjustment of image projection by translation in image coordinates is accurate over each of the sites' ground footprints due to the high relative accuracy achieved by commercial satellite ground processing. For example, an adjustment of image offset typically achieves projection accuracy of less than one pixel over the extent of a typical World View image of 40×100 kilometers. Larger extents, such as mapping strips, may require a full affine transformation of image coordinates, which is still a linear operation.

Before identifying the coordinate systems and transformations used to set up the error covariance model, we first note the following metadata that typically accompanies the satellite imagery:

- rational polynomial coefficient (RPC) model for projecting ground points to image points
- the origin point, R_o on the Earth surface in WGS84 coordinates, (λ_o, ϕ_o, h_o) , where h_o is some nominal height above the WGS84 Ellipsoid.
- satellite azimuth az and elevation eE during image collection
- altitude of the satellite orbit, H_s
- inclination angle θ of the orbit with respect to the equator

There are several coordinate systems involved in computing the pose covariance matrix and its effect on the intersection of sensor rays in the formation of multi-image geometry, as shown in Fig. 7. The relationship between the satellite sensor and a local coordinate system on the ground is shown in a). A tangent plane to the Earth's surface is defined at an origin point, R_o , shown in the figure. An East-North-up (enu) coordinate system is constructed at R_o . The North (N_o) and up (up_o) axes are shown in the figure. The orientation of the satellite with respect to the enu coordinate system is defined by the azimuth (az) and elevation angles (eE) shown in the figure. A unit vector \hat{u}_s is computed from the azimuth and elevation angles and is expressed in the Earth-centered, Earth-fixed (ecf) coordinate system. The position of the satellite R_s in the Earth-centered, Earth-fixed (ecf) coordinate system is computed by intersecting a vector of unknown length starting at R_o and pointing along \hat{u}_s with a geocentric sphere whose radius is the sum of the known height of the orbit H_s and the nominal radius of the Earth R_e .

Fig. 7b) shows the in-track, cross-track, radial (icr) coordinate system. The radial vector corresponds to the geocentric radius vector of the satellite (R_s) and is not aligned with the \hat{u}_s vector except for the special case of nadir imaging. The tangent vector to the orbit (in-track) is computed from the known inclination angle, θ , of the satellite which corresponds to the ground track angle at the equator; the current algorithm does correct for the change in the ground track angle as latitudes increase in magnitude towards the poles. The cross-track coordinate vector is defined by the cross product of the radial and in-track vectors.

Fig. 7c) defines the sensor coordinate system. A satellite can change its orientation during image collection to point off-nadir and so the sensor coordinate system is generally not aligned with the icr coordinate system. The sensor coordinates U, V, W are defined such that W is aligned with the view direction of the optical system and pointing away from the Earth. The V axis is perpendicular to W and parallel to the scan direction of the image. The unit vector U is defined as the cross-product of V and W , which completes the sensor coordinate system. The angles ω, ϕ, κ represent variations with respect to the measured attitude of the satellite and are a key source of pointing inaccuracy.

B. Pose covariance

A physical basis for pose uncertainty is now defined with respect to errors in satellite position and orientation (pose), so-called *adjustable parameters*. The error in satellite pose is defined as,

$$d\phi = \begin{bmatrix} dI \\ dC \\ dR \\ \omega_s \\ \phi_s \\ \kappa_s \end{bmatrix}, \quad (14)$$

where (dI, dC, dR) are the in-track, cross-track and radial errors in satellite position and $(\omega_s, \phi_s, \kappa_s)$ are small angle errors in satellite sensor attitude as shown in Fig. 7c). The position error vector can be transformed into the sensor coordinate frame according to,

$$\begin{bmatrix} dU \\ dV \\ dW \end{bmatrix} = T_{ecf \rightarrow sensor} T_{icr \rightarrow ecf} \begin{bmatrix} dI \\ dC \\ dR \end{bmatrix}, \quad (15)$$

where $T_{ecf \rightarrow sensor}$ is the rotation from ecf coordinates to the sensor coordinate system (sX, sY, sZ), and $T_{icr \rightarrow ecf}$ is the rotation from the satellite in-track, cross-track and radial coordinate system to ecf coordinates. See Fig. 7c). This transformation is incorporated later in the Jacobian that maps the pose errors into errors in image ray displacements.

The pose covariance matrix, denoted as S_ϕ , is a $6n \times 6n$ matrix, where n is the number of images. For a single image and for the WorldView3 satellite, the covariance matrix is,

$$S_\phi = \begin{bmatrix} 0.5m^2 & & & & & \\ & 0.5m^2 & & & & \\ & & 0.5m^2 & & & \\ & & & 8 \times 10^{-12} rad^2 & & \\ & & & & 8 \times 10^{-12} rad^2 & \\ & & & & & 16 \times 10^{-12} rad^2 \end{bmatrix}$$

As an illustration, for a nadir image the corresponding variance in ground position is,

$$\sigma_{xx} = 620000^2 \times 8 \times 10^{-12} + 0.5 = 3.6m^2$$

where the orbital altitude is 620 km.

The pose covariance matrix does not explicitly contain terms to model jitter, which is applicable to sensor attitude, but not position due to the stable orbit. The magnitude of the jitter component of attitude error is at least an order of magnitude smaller than that of the global pointing error, as demonstrated by the bundle adjustment residuals, typically less than 0.5 pixel in magnitude, and therefore already accommodated by the assigned variance.

It is also possible for sensor pose errors of two or more images to be correlated. The correlation is due to the persistence of error states for short periods of time along the same orbital pass. Sensor position and attitude errors for nearby images are highly correlated; e.g., a correlation coefficient on the order of $\rho = 0.8$. Pose errors are typically modeled as completely uncorrelated between images acquired from separate orbital passes even if acquired from the same sensor.

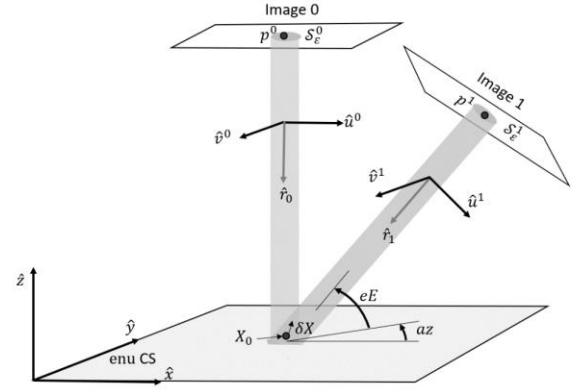


Fig. 8 The intersection of two rays. The ray direction unit vectors are \hat{u}^0 and \hat{u}^1 with origin points, p^0 and p^1 , and covariance matrices S_ϵ^0 and S_ϵ^1 respectively.

V. PROPAGATING SATELLITE POSE COVARIANCE

A. Ray covariance

The intersection of two sensor rays is illustrated in Fig. 8. The sensor ray is shown as a semitransparent cylinder indicating that only displacements of a ray orthogonal to the ray affect the intersection point $X = X_0 + dX$. The ray displacements are due to pose errors during image collection. The covariance of ray origin p^i is defined with respect to the unit vectors, \hat{u}^i and \hat{v}^i , which are the unit sensor coordinate vectors $s\hat{X}$ and $s\hat{Y}$ transformed into the local enu Cartesian coordinate frame, denoted as $(\hat{x}, \hat{y}, \hat{z})$ in Fig. 8. \hat{u}^i and \hat{v}^i are orthogonal to each other and to the ray direction, \hat{r}^i . The perturbed ray origin is given by, $\tilde{p}^i = p^i + \epsilon_u^i \hat{u}^i + \epsilon_v^i \hat{v}^i$, where $(\epsilon_u^i, \epsilon_v^i)$ are random variables representing the ray displacement errors due to errors in sensor pose. For n images the collective error vector is denoted as,

$$\mathcal{E}_s = \begin{bmatrix} \varepsilon_u^0 \\ \varepsilon_v^0 \\ \varepsilon_u^1 \\ \varepsilon_v^1 \\ \vdots \\ \varepsilon_u^{n-1} \\ \varepsilon_v^{n-1} \end{bmatrix} \quad (16)$$

and the ray covariance matrix $\mathcal{S}_{\varepsilon_s} = \langle \mathcal{E}_s \mathcal{E}_s^T \rangle$.

In order to relate the satellite pose error covariance to the ray covariance it is necessary to construct a Jacobian derivative matrix. For a single sensor the orthogonal ray displacements are,

$$\begin{bmatrix} \varepsilon_u \\ \varepsilon_v \end{bmatrix} = \begin{bmatrix} (dU(dI, dC, dR) + |R|\phi_s) \\ (dV(dI, dC, dR) - |R|\omega_s) \end{bmatrix} \quad (17)$$

Note that displacements along the ray and rotations about the ray, κ_s do not affect the intersection point, X . The elements involving κ_s of the pose covariance matrix, \mathcal{S}_{ϕ} , can be dropped giving a $5n \times 5n$ matrix. The full $2n \times 5n$ Jacobian matrix is,

$$J = \frac{\partial \mathcal{E}_s}{\partial \phi} = \begin{bmatrix} dU_I^0 & dU_C^0 & dU_R^0 & 0 & |R^0| & \dots & 0 & 0 & 0 & 0 & 0 \\ dV_I^0 & dV_C^0 & dV_R^0 & -|R^0| & 0 & \dots & 0 & 0 & 0 & 0 & 0 \\ \vdots & \vdots & \vdots & \vdots & \vdots & \ddots & \vdots & \vdots & \vdots & \vdots & \vdots \\ 0 & 0 & 0 & 0 & 0 & \dots & dU_I^{n-1} & dU_C^{n-1} & dU_R^{n-1} & 0 & |R^{n-1}| \\ 0 & 0 & 0 & 0 & 0 & \dots & dV_I^{n-1} & dV_C^{n-1} & dV_R^{n-1} & -|R^{n-1}| & 0 \end{bmatrix} \quad (18)$$

The $2n \times 2n$ ray covariance matrix is then $\mathcal{S}_{\varepsilon_s} = J \mathcal{S}_{\phi} J^T$. Note that J is block diagonal, i.e., $\frac{\partial \varepsilon_i^k}{\partial \phi^j} = 0, i \neq j$.

B. Ray intersection point covariance

The ray intersection algorithm is based on a well-known linear algorithm where the solution minimizes the sum of squared distances from each ray to the intersection point [17]. A projection matrix $P_{r,i} = (I - \hat{r}^i (\hat{r}^i)^T)$ is defined that produces the component of a vector that is orthogonal to ray \hat{r}^i . That is, $d_{\perp}^i(X) = P_{r,i}(p^i - X)$, where $d_{\perp}^i(X)$ is the vector component of $(p^i - X)$ orthogonal to ray \hat{r}^i . The linear algorithm minimizes the sum of squared perpendicular distances, $D(X) = (\delta_{\perp}(X))^T \delta_{\perp}(X)$ where,

$$\delta_{\perp}(X) = \begin{bmatrix} d_{\perp}^0(X) \\ d_{\perp}^1(X) \\ \vdots \\ d_{\perp}^{n-1}(X) \end{bmatrix} \quad (19)$$

The solution is found by setting the derivative of $D(X)$ to zero.

$$\frac{\partial D(X)}{\partial X} = -2 \sum_i P_{r,i}(p^i - X) = 0$$

$$A = \sum_i P_{r,i}, AX = \sum_i P_{r,i} p^i, X = A^{-1} \sum_i P_{r,i} p^i \quad (20)$$

The solution in Eq. (20) can be seen as one that assigns the average orthogonal components of X to the average orthogonal components of the ray origins.

The algorithm can be extended to take into account the error covariance of satellite pose by extracting the 2-d scalar

components of the ray displacements in the coordinate system of the plane orthogonal to the ray. Define a 2×3 projection matrix $\pi_{\perp}^i = \begin{bmatrix} (\hat{u}^i)^T \\ (\hat{v}^i)^T \end{bmatrix}$ that maps a 3-d vector into the corresponding 2-d plane coordinates. The 2-d vector $d_{\perp}^i(X) = \pi_{\perp}^i(p^i - X)$ lies in the orthogonal plane as shown in Fig. 9.

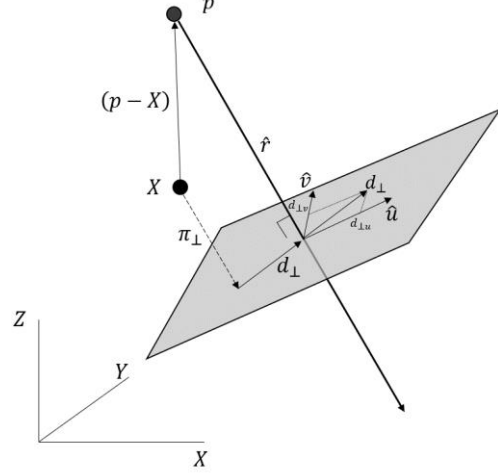


Fig. 9 The projection of a 3-d vector into 2-d coordinates of the plane orthogonal to ray r .

As mentioned above, errors in satellite pose only affect the ray origin so $\frac{\partial d_{\perp}^i(X)}{\partial \varepsilon_u^i, \varepsilon_v^i} = \begin{bmatrix} 1 & 0 \\ 0 & 1 \end{bmatrix}$. Let the $2n \times 3$ matrix, Π , be defined as,

$$\Pi = \begin{bmatrix} \pi_{\perp}^0 \\ \pi_{\perp}^1 \\ \vdots \\ \pi_{\perp}^{n-1} \end{bmatrix} \quad (21)$$

Let the $3n \times 1$ matrix, \mathcal{R} , represent the set of vectors from the intersection point, X , to the ray origins, p^i .

$$\mathcal{R}(X) = \begin{bmatrix} (p^0 - X) \\ (p^1 - X) \\ \vdots \\ (p^{n-1} - X) \end{bmatrix} \quad (22)$$

The projection operation $\Pi \Vdash \mathcal{R}(X)$ produces a $2n \times 1$ matrix and denotes the simultaneous projection of each 3-d vector element of $\mathcal{R}(X)$, $(p^i - X)$, by the corresponding π_{\perp}^i projection matrix. A weighted least squares solution can now be formulated where the scalar quantity to be minimized is,

$$\tilde{D}(X) = (\Pi \Vdash \mathcal{R}(X))^T (\mathcal{S}_{\varepsilon})^{-1} \Pi \Vdash \mathcal{R}(X) = \mathcal{R}^T(X) \Pi^T (\mathcal{S}_{\varepsilon})^{-1} \Pi \Vdash \mathcal{R}(X). \quad (23)$$

The evidence from each 2-d orthogonal displacement is weighted by the inverse ray covariance matrix so that more accurate ray positions play a greater role in the determination of the intersection point. The solution for the intersection point proceeds as follows.

$$\frac{\partial \tilde{D}(X)}{\partial X} = -2 \Pi^T (\mathcal{S}_{\varepsilon_s})^{-1} \Pi \Vdash \mathcal{R}(X) = 0 \quad (24)$$

Define the vector of ray origins as \mathbb{P} , a $3n \times 1$ matrix,

$$\mathbb{P} = \begin{bmatrix} p^0 \\ p^1 \\ \vdots \\ p^{n-1} \end{bmatrix} \quad (25)$$

and \mathbb{X} as a $3n \times 1$ matrix of copies of the unknown intersection point,

$$\mathbb{X} = \begin{bmatrix} X \\ X \\ \vdots \\ X \end{bmatrix} = \begin{bmatrix} I_{3 \times 3} \\ I_{3 \times 3} \\ \vdots \\ I_{3 \times 3} \end{bmatrix} X \quad (26)$$

With these definitions, the derivative of the weighted total squared error is

$$\frac{\partial \tilde{D}(X)}{\partial X} = -2\Pi^T(\mathcal{S}_{\mathcal{E}_s})^{-1}\Pi \left(\mathbb{P} - \begin{bmatrix} I_{3 \times 3} \\ I_{3 \times 3} \\ \vdots \\ I_{3 \times 3} \end{bmatrix} X \right) = 0 \quad (27)$$

Let $\tilde{\mathcal{A}} = \Pi^T(\mathcal{S}_{\mathcal{E}_s})^{-1}\Pi$, and it follows that,

$$X = \tilde{\mathcal{A}}^{-1} \left(\begin{bmatrix} [\hat{u}^0 & \hat{v}^0] & [\hat{u}^1 & \hat{v}^1] & \dots & [\hat{u}^{n-1} & \hat{v}^{n-1}] \end{bmatrix} (\mathcal{S}_{\mathcal{E}_s})^{-1} \begin{bmatrix} (\hat{u}^0)^T \\ (\hat{v}^0)^T \\ (\hat{u}^1)^T \\ (\hat{v}^1)^T \\ \vdots \\ (\hat{u}^{n-1})^T \\ (\hat{v}^{n-1})^T \end{bmatrix} \right) \Pi \begin{bmatrix} p^0 \\ p^1 \\ \vdots \\ p^{n-1} \end{bmatrix} \quad (28)$$

It can be shown that the covariance of the 3-d intersection point coordinates is $\tilde{\mathcal{A}}^{-1} = P$ (see Eq. 3), and is the same as the Multi-Image Geo-positioning (MIG) solution but in this case is a natural byproduct of the ray intersection solution. Eq. (28) is the key result where the computation of the multi-ray intersection point and error propagation of satellite pose covariance occurs at the same time.

To illustrate the benefit of the covariance-weighted solution an example set of 17 WorldView3 images was selected. The image set is composed of three orbital passes and the pose covariance matrix contains correlations between satellite attitude errors within each pass. The spread of view directions for the orbital passes is shown in Fig. 16. The resulting 85×85 covariance matrix defines a multi-normal distribution for satellite pose errors. Samples are generated from this distribution to produce an ensemble of perturbed ray positions. The 17 rays for each sample are intersected using the original linear algorithm without covariance weighting. The scatter of the resulting 3-d intersection points for 100,000 samples is shown in Fig. 10.

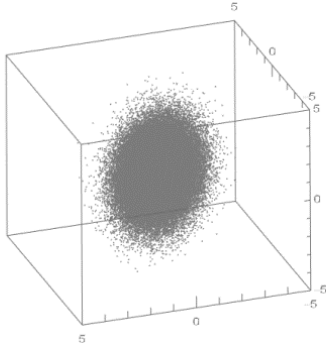


Fig. 10 The scatter of 3-d intersection points due to satellite pose error. Plot units are meters.

The covariance matrix of these samples \mathcal{S}_{samp} is compared with the covariance matrix for X as determined by $\tilde{\mathcal{A}}^{-1}$. The uncertainty ellipsoid with covariance weighting has half the volume of that without covariance weighting, demonstrating much improved accuracy.

The global position of the ray intersection point is therefore significantly more accurate than its predicted position from any single image. The intersection point can be used to correct the global bias of each image by projecting the point into the image and translating the image coordinate system to align the image location of the intersection point with the location of the projected 3-d point. In addition, this alignment procedure co-registers the images to sub-pixel relative accuracy as is needed for stereo reconstruction. Additional transformational degrees of freedom such as rotation and scale are not needed, as it is observed that minimum elevation errors across the span of the datasets do not exceed 0.2m.

VI. LOCAL ERRORS

A. DSM point grids

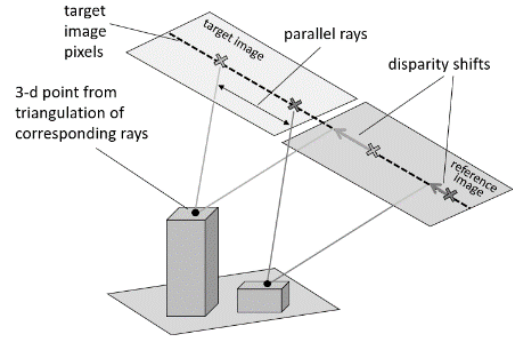


Fig. 11 Disparity shifts between the target and reference images of a rectified stereo pair results in 3-d points constructed by intersecting rays.

Each stereo pair generates a disparity image found by the SGM algorithm. For details of the algorithm, see Hirschmüller [6]. The disparity values represent the difference in image locations for corresponding surface positions along rows of the epipolar-rectified left and right image. As shown in Fig. 11, surfaces closer to the sensor produce larger disparity shifts than more remote surfaces. Note that the camera rays for a given image are parallel in accordance with the affine projection model, see Eq. (8) and Fig. 6. 3-d points are found by intersecting the pair of rays defined by the disparity image. A typical stereo pair will generate more than one million 3-d points per tile that densely cover the observed scene surfaces.

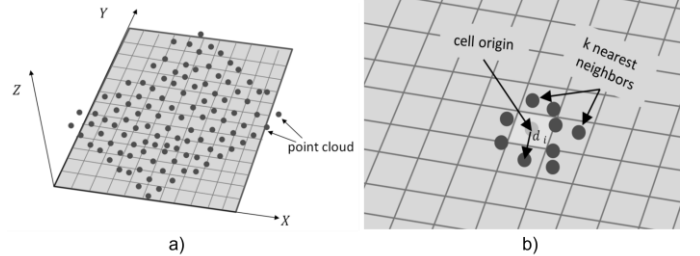


Fig. 12 Associating a stereo point cloud with the DSM pixel grid. a) The dense point cloud is not generally aligned with the DSM pixel grid. b) The elevation for each pixel cell is found by interpolating nearby values.

B. Horizontal error

The point clouds for each stereo pair are fused together by associating the points with DSM pixel domains resulting in a set of elevation values for each pixel location as shown in

Fig. 12. The stereo image pixel space and the 3-d DSM grid coordinate space will necessarily be different and so the dense point cloud generated by a stereo pair will not be aligned with DSM bins, as shown in Fig. 12a). A DSM pixel bin (i, j) is associated with a set of k neighbors $N_{i,j}$ that lie within a specified radius r of the bin center (x_o, y_o) .

That is $N_{i,j} = \{X_r | d_r \leq r, |N_{i,j}| \leq k\}$,

$d_r = \sqrt{(x_r - x_o)^2 + (y_r - y_o)^2}$. As mentioned in the introduction, each point X_r is associated with a probability P_r based on the consistency in 3-d point locations between the forward and reverse order of the stereo pair. The elevation at a bin is determined by a weighted least squares algorithm where the weight w_r of a point is, $w_r = \frac{P_r}{d_r}$. The point probabilities are also interpolated to assign a probability to the bin itself, $\bar{P}_q(i, j)$, with respect to a given stereo pair q .

In general, the points that are assigned to a bin are scattered randomly about the bin center. Thus, the elevation assigned to the bin will not correspond to the actual world surface elevation at the bin geographic coordinates. The elevation is more accurately associated with the location of the neighborhood centroid but with a standard deviation related to the scatter of the point locations within the neighborhood. The horizontal squared displacement at bin (i, j) for stereo pair q is given by,

$$\sigma_{hq}^2(i, j) = \frac{1}{\sum_r P_r(i, j)} \sum_r P_r(i, j) d_r^2(i, j) \quad (29)$$

It is necessary to combine these horizontal displacement values from each stereo pair to obtain the final displacement for the fused DSM. The fused horizontal displacement is given by,

$$\bar{\sigma}_h^2(i, j) = \frac{1}{\sum_q \bar{P}_q(i, j)} \sum_q \bar{P}_q(i, j) \sigma_{hq}^2(i, j) \quad (30)$$

An example of the fused horizontal root mean squared displacement is shown in Fig. 14. As might be expected, the displacement increases markedly near step boundaries at building edges. There are also somewhat higher displacements on some sloped surfaces and near slope discontinuities.

C. Vertical error

The fusion of multiple elevation values at each DSM pixel bin is achieved by forming a weighted set of elevation values that are consistent with each other. In this context a set

$$C(i, j) = \{z_q(i, j) | \text{abs}(z_q(i, j) - z_l(i, j)) < \text{tol}\} \quad (31)$$

is consistent if all values are within a tolerance of an initial seed $z_l(i, j)$ used to start the formation of the set. The fusion process tries all elevation values at (i, j) as seeds and selects the consensus set with the largest expected number of members $\langle N_C \rangle$, where $\langle N_C \rangle = \sum_{q \in C} P_q$ and P_q is the probability of $z_q(i, j)$.

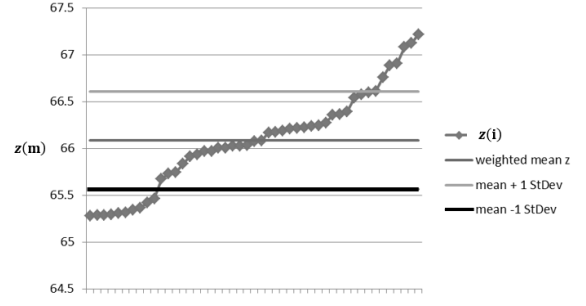


Fig. 13 An example consensus set. The mean elevation and $\pm 1\sigma$ bounds are shown.

An example of a consensus set of elevations is shown in Fig. 13. The elevations are sorted to better illustrate the range of values. The elevation value for the bin is assigned the weighted mean of the elevations,

$$\bar{z}(i, j) = \frac{1}{\sum_{q \in C} P_q} \sum_{q \in C} P_q z(i, j) \quad (32)$$

with standard deviation,

$$\sigma_z(i, j) = \sqrt{\frac{1}{\sum_{q \in C} P_q} \sum_{q \in C} P_q (z(i, j) - \bar{z}(i, j))^2} \quad (33)$$

The $\pm 1\sigma_z$ limits about the mean are also shown in Fig. 13.

This analysis provides a prediction of the expected local vertical errors at each pixel of the DSM. An example profile scan of a series of solar panels is shown in Fig. 19. Note that the DSM elevation error is within the 90% linear error bounds except for a few isolated points.

VII. EXPERIMENTAL RESULTS

A. Global Error

a) Buenos Aires, Argentina

A site in Buenos Aires, Argentina, (34.4894120S, 58.5859220W), is shown in Fig. 15. The feature correspondence track has 29 WorldView3 images as shown in Fig. 15a). Four of the images are collected on the same orbital pass as indicated by orange collinear points. All five of the pose components of the four images are correlated with $\rho = 0.8$. The DSM must be translated by $t_{bias} = (0.59, 0.49, 0.98)$ m to align with a LiDAR DSM covering the same region. The LiDAR DSM has 0.5m spacing and the DSM, Fig. 15b), has 0.3m spacing. The alignment is carried out by converting both to point clouds and minimizing the distances to the nearest points.

The effect of attitude correlation on four out of 29 images was investigated by setting $\rho = 0.0$ and reprocessing the DSM with the modified geo-registration. The registration algorithm that aligns the DSM and LiDAR point clouds has a standard deviation over repeated registration trials of about 0.1m for translation components. Thus, no significant difference is observed with $t_{bias, \rho=0} = (0.485, 0.547, 1.012)$ m. Finally, the $2n \times 2n$ covariance matrix of the ray displacements was set to the identity matrix to observe the bias without considering view obliquity effects. Again, no significant change in bias was observed, $t_{bias, cov=I} = (0.441, 0.563, 0.980)$ m. This result is to be expected since the pose variances are approximately the same for each image only differing with

respect to the degree of view direction obliquity. If pose variances are identical for each image, the common variance value can be factored out of the expression for squared perpendicular error and the result reverts to ray intersection without covariance weighting.

b) *Wright Patterson AFB, Dayton, OH*

The WorldView3 image data for the Wright Patterson AFB experiment is an unusual case where most of the images are in disjoint subsets that are collected within seconds of each other on single orbits. There are three single-pass groups as indicated by the orange, red and green points in Fig. 16 a). There will be a high degree of correlation between the position and attitude errors of the satellite pose for each of the orbital passes. Accounting for these correlations significantly reduces the resulting horizontal (East-West) bias in the 3-d ray intersection point for the 19-image track.

The 3-d covariance matrix with pose correlation coefficient $\rho = 0.8$ is

$$P = \tilde{\mathcal{A}}^{-1} = \begin{bmatrix} 0.501598 & 0.004349 & -0.02467 \\ 0.004349 & 0.526126 & -0.02467 \\ -0.02467 & -0.02467 & 1.220778 \end{bmatrix}.$$

The required translation to align the DSM with LiDAR ground truth is $t_{bias} = 1.691, -0.385, 0.540$. The resulting bias without considering pose covariance is

$$t_{bias, cov. = I} = 2.079, -0.414, -0.153,$$

which exceeds the predicted 90% bias error ellipsoid in Fig. 16 c), d). This experiment shows that correctly modeling pose correlation on single pass collection sequences decreased the horizontal error component of the bias, while it increased the vertical bias component.

c) *Richmond, Virginia*

Table 1 Distribution of sensor types for the Richmond dataset

Satellite Platform	Number	Position Std. Dev.	Attitude Std. Dev.
GeoEye-1	12	0.7071	2×10^{-6}
QuickBird	3	1.0	23.203×10^{-6}
WorldView1	1	0.7071	3.742×10^{-6}
WorldView2	23	0.7071	2.83×10^{-6}
WorldView3	5	0.7071	2.83×10^{-6}

There is a heterogeneous mix of satellite sensor types in this example. The correspondence track has 44 images with the population of satellite platforms shown in Table 1. The significant difference in error covariance among the different sensor platforms illustrates the benefit of applying error propagation in the computation of the 3-d track intersection point and the resulting DSM geographic bias. The original DSM geo-registration algorithm using non-linear bundle adjustment without covariance weighting had produced a bias of $t_{bias, BA} = (0.180, -1.203, 1.629)$ compared to the new $t_{bias, cov.} = (0.583, -0.696, 0.349)$ with covariance weighting and the linear least squares ray intersection algorithm. The effect of covariance weighting is very significant – the magnitude of the bias translation vector with

covariance weighting is approximately one half that produced by the original bundle adjustment algorithm.

The significant difference in error covariance among the different sensor platforms emphasizes the benefit of applying error propagation in the computation of the 3-d track intersection point and the resulting DSM geographic bias.

B. Local Error

a) *Vertical Error*

The accuracy of vertical error prediction for a region of the University of California at San Diego DSM is shown in Fig. 18. In the experiments to follow, the DSM and the LiDAR GEOTIFFs are converted to 3-d point clouds and the closest in (x, y) position LiDAR point to a given DSM point is considered to define the ground truth elevation. The ground truth and DSM elevations are shown as colorized images in Fig. 18 a) and b). The scales indicate the range of elevations. The difference in elevations, $(z - \bar{z})$, is shown in Fig. 18 c).

Note that there are areas of large negative value (blue) due to differences in vegetation between the DSM and the LiDAR. These differences are due to the relatively long time span, 2014 – 2019, of image collections used to construct the DSM and the nearly instantaneous collection time of the LiDAR. The vegetation, as well as transient vehicles, and parking lot construction, varied over the five-year period but all of these elevation changes are fused into a single height estimate in the final DSM. Persistent structures are accurately represented in the DSM and appear as red in Fig. 18c).

Other than the vegetation areas, error prediction is consistent with the observed elevation differences except at building roof boundaries. Note the fine green linear edges in Fig. 18e) exhibiting higher normalized distances. As was noted earlier, these differences may often be accounted for by the horizontal error tolerance that permits a range of ground truth positions to be matched with a single DSM elevation. This point will be examined later in the discussion on horizontal error.

The accuracy of vertical error prediction for a sloped surface from the same region can also be conveyed by plots where the elevation values lying along a line of DSM samples is compared to LiDAR at the same sample locations. An example from UCSD is shown in. In this example the observed differences are mostly bounded by the LE90 limits and a large fraction of the points are well within 1σ limits.

a) *Horizontal Error*

It is difficult to measure horizontal error directly for several reasons:

- the LiDAR DSM ground truth pixel spacing, 0.5m, is on the order of the horizontal error
- changes in elevation where horizontal position error might be estimated occur only at a small fraction of the surface grid cells

Another approach is to augment vertical error bounds evaluation by examining the population of ground truth elevations within a neighborhood defined by the horizontal error CE90 radius. The closest ground truth elevation within the neighborhood to the DSM test point can be taken as the

true value. The horizontal neighborhood radius r_{h90} is defined as,

$$r_{h90}(i, j) = 2.146 \bar{\sigma}_h(i, j) + \frac{1}{\sqrt{2}} s_{LiDAR}$$

, where s_{LiDAR} is the LiDAR cell spacing (0.5m). The term $\frac{1}{\sqrt{2}} s_{LiDAR}$ defines the diagonal distance to the center of a LiDAR pixel so that ground truth neighbors just touching the horizontal position uncertainty boundary are included. The term 2.146 accounts for the circular error 90% uncertainty due to normally distributed random point scatter. This tolerance is seen to significantly reduce vertical error near step boundaries.

VIII. CONCLUSIONS AND FUTURE WORK

The error prediction results just described provide a comprehensive approach to characterizing the expected uncertainty of DSM products. It has been shown that attitude correlation for satellite image collections along the same orbital pass can have a very significant effect on the accuracy of DSM geo-positioning. In the Wright Patterson AFB experiment of Fig. 16, the geo-positioning bias in the Easterly direction is reduced to be within the 90% ellipsoidal bounds, while error prediction fails without accounting for such correlation. In the case of the Richmond, VA experiment, the application of satellite pose covariance weighting for a mix of satellite sensor types reduces the geo-positioning bias vector magnitude by one half, compared to the unweighted solution, while at the same time producing a predicted 90% ellipsoid that properly bounds the actual observed bias.

The approach to local elevation error prediction enables the specification of uncertainty at every DSM sample. These predictions bound the observed error with respect to registered LiDAR within one standard deviation at most surface locations except for surface step discontinuities and vegetation changes due to the different times of acquisitions of the datasets. The errors at such boundaries are significantly higher than predicted by the standard deviation computed from DSM bin elevation consensus sets and, in some cases, do not lie within the LE90 bounds defined by $1.64\sigma_z$. These discrepancies are due largely to errors in the horizontal location of the step boundary and to the discrete spacing (0.5m) of the LiDAR ground truth surface points. If the horizontal tolerance is relaxed to include a neighborhood of ground truth points then the elevation errors at step boundaries are dramatically reduced with most surface points lying within the predicted bounds.

The predicted error is significantly higher than observed error for smooth surfaces such as building roofs. Future investigations will consider the measured DSM surface roughness within a neighborhood around each DSM grid cell in forming a tighter bound on predicted local error over smooth surfaces. That is, the error prediction will be conditioned on local surface geometry.

Future experiments will focus on expanding the number and type of sites to validate both global bias terms and local error contributions to predicted geolocation uncertainty. The use of satellite pose covariance weighting to improve DSM geographic placement accuracy depends on the accuracy of the specified position and attitude variances. It is possible to

estimate these variances from correspondence tracks that have a small absolute geographic bias relative to the satellite pointing error. Various points on the satellite orbit will be examined to determine pose error variance estimates and their variation with respect to satellite position and attitude.

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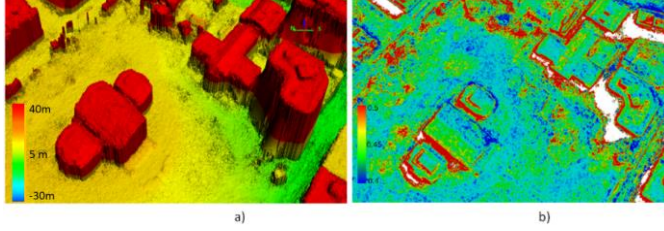


Fig. 14 The horizontal error for a DSM of Richmond, VA with 0.3 m pixel spacing and 100 fused stereo pairs. a) A region of the DSM. b) The horizontal root mean squared displacement $\bar{\sigma}_h$. Red indicates a larger displacement, blue smaller. The white pixels are undefined due to shadows. The peak of the distribution of $\bar{\sigma}_h$ is 0.45m, slightly larger than the grid spacing of 0.3m.

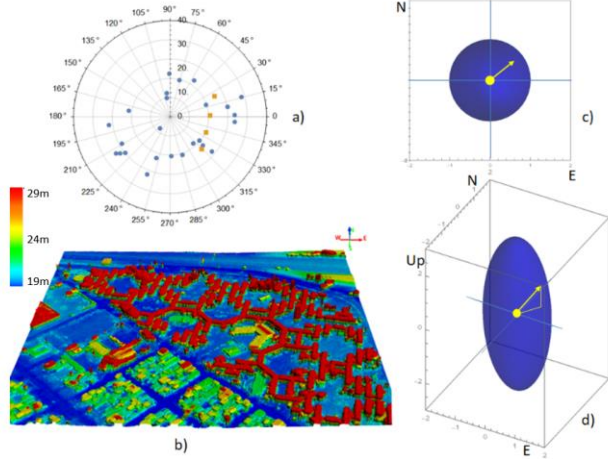


Fig. 15 Global error analysis for Buenos Aires. a) The azimuth and elevation of the satellite view directions (standard spherical coordinates). The orange squares indicate a single pass. b) The DSM for the region of interest. c), d) Views of the 90% error ellipsoid in enu coordinates (m). The yellow arrow indicates geographic bias in the placement of the DSM.

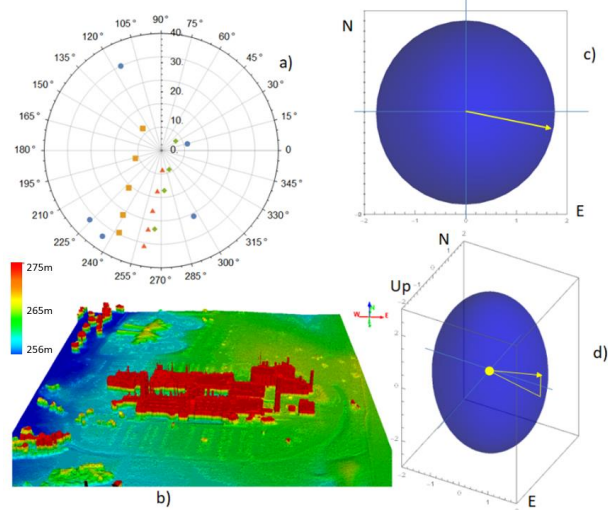


Fig. 16 Global error analysis for Wright Patterson AFB. a) The azimuth and elevation of the satellite view directions (standard spherical coordinates). There are 19 images with three single pass groups (orange, red, green). b) The DSM for the region of interest, 0.3m pixel spacing. c), d) Views of the 90% error ellipsoid in enu coordinates (m). The yellow arrow indicates geographic bias in the placement of the DSM.

bias in the placement of the DSM.

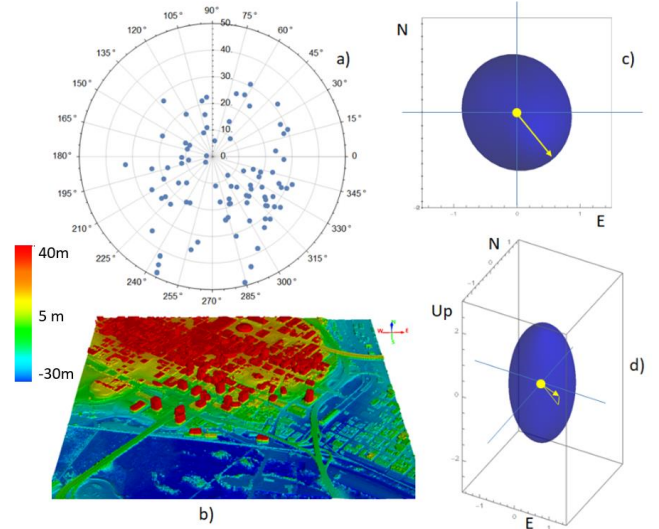


Fig. 17 Global error analysis for the Richmond site. a) The azimuth and elevation of the satellite view directions (standard spherical coordinates). There are 44 images in the single correspondence track. b) The DSM for the region of interest, 0.3m pixel.

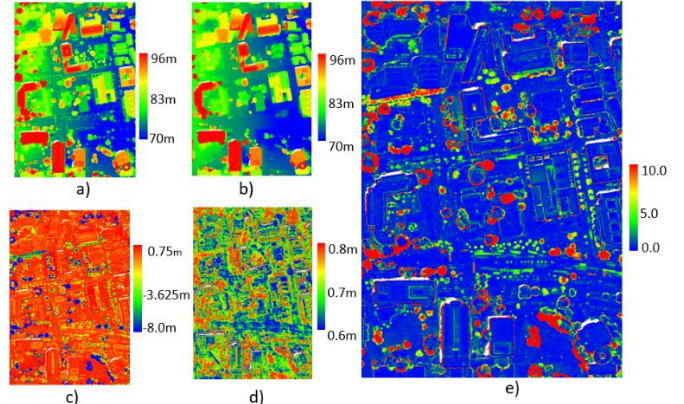


Fig. 18 Vertical error at UCSD. a) LiDAR ground truth (0.5m). b) DSM (0.3m). c) $(z - \bar{z})$, where \bar{z} is the LiDAR ground truth elevation. d) The vertical error standard deviation σ_z . e) Normalized distance, $(z - \bar{z})/\sigma_z$.

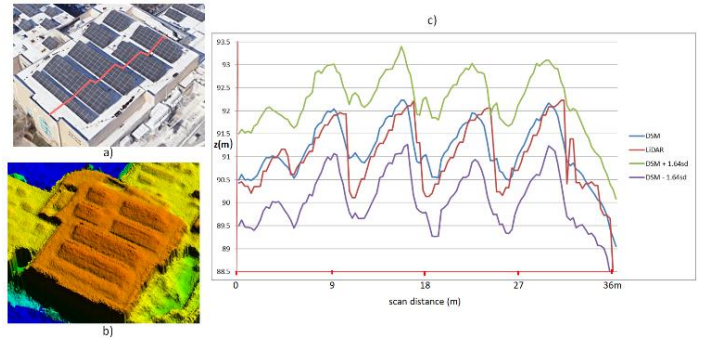


Fig. 19 A cross section plot comparing DSM elevations with LiDAR. a) A view of the scan path sketched on Google Earth (Copyright Google Earth) b) A view of the DSM surface for the same region. c) A line plot of the DSM values (blue) compared to LiDAR (red). Also shown are the 90% linear error bounds, green, purple ($\pm 1.64\sigma$). (Note that σ is a function of position and so the bounds contours are not just constant offsets to the DSM elevations.)